



LUVOIR ARCHITECTURE “A” ENGINEERING STATUS

**Presented to:
The LUVOIR STDT**

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February 21, 2017**

We will study two architectures in depth...

◎ Architecture A (first half of 2017)

- 15-m diameter aperture
- Five instrument bays:
 - Optical / NIR Coronagraph (A)
 - UV Multi-object Spectrograph (“LUMOS”)
 - High-definition Imager (will also perform guiding / wavefront sensing)
 - *UV Spectro-polarimeter (CNES Contributed)*
 - *Empty Bay for future expansion / contribution*

◎ Architecture B (late 2017 into 2018)

- ~9-m diameter aperture
- Three instrument bays:
 - Optical / NIR Coronagraph (B)
 - UV Multi-object Spectrograph (“LUMOS”)
 - Optical / NIR Multi-resolution Spectrograph
 - Will need to include guiding and wavefront sensing capabilities

Study Schedule (2017):

- ✓ **1/17–24** – Telescope Instrument Design Lab (IDL)
 - Pre-work 1/10
- ✓ **2/6–10** – HDI IDL
 - Pre-work 1/31
- ⊙ **3/20–24** – Coronagraph IDL
 - Pre-work 3/14
- ⊙ **5/15–19** – LUMOS IDL
 - Pre-work 5/9
- ⊙ **6/7–13** – Instrument Accommodation & Δ Telescope IDL
 - Pre-work 6/1
- ⊙ **7/10–14** – LUVOIR “A” Mission Design Lab (MDL)
- ⊙ June – Dec.: Prepare Interim Report on Architecture A
- ⊙ Sept.: Kick-off Architecture B IDLs

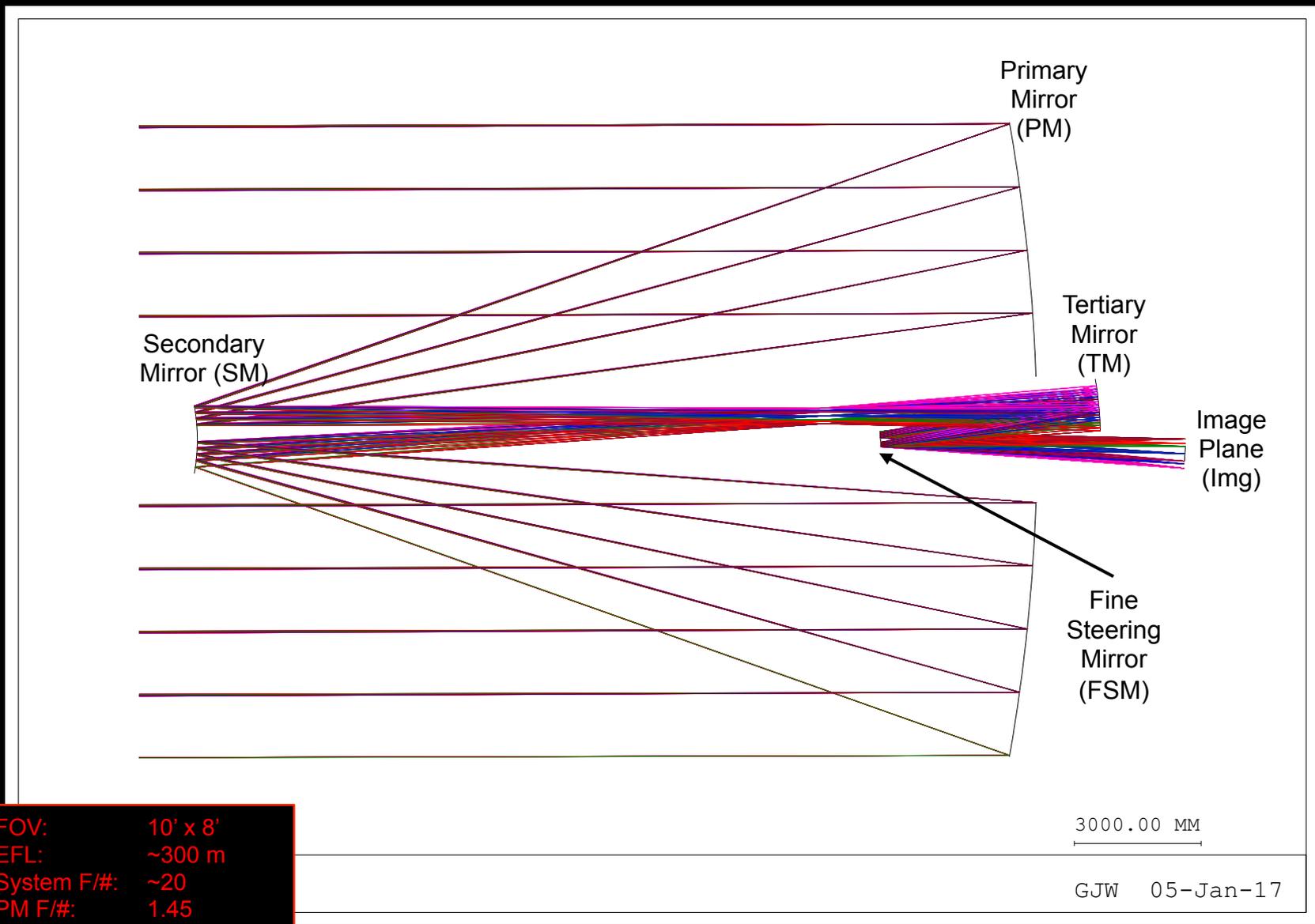
Telescope Design

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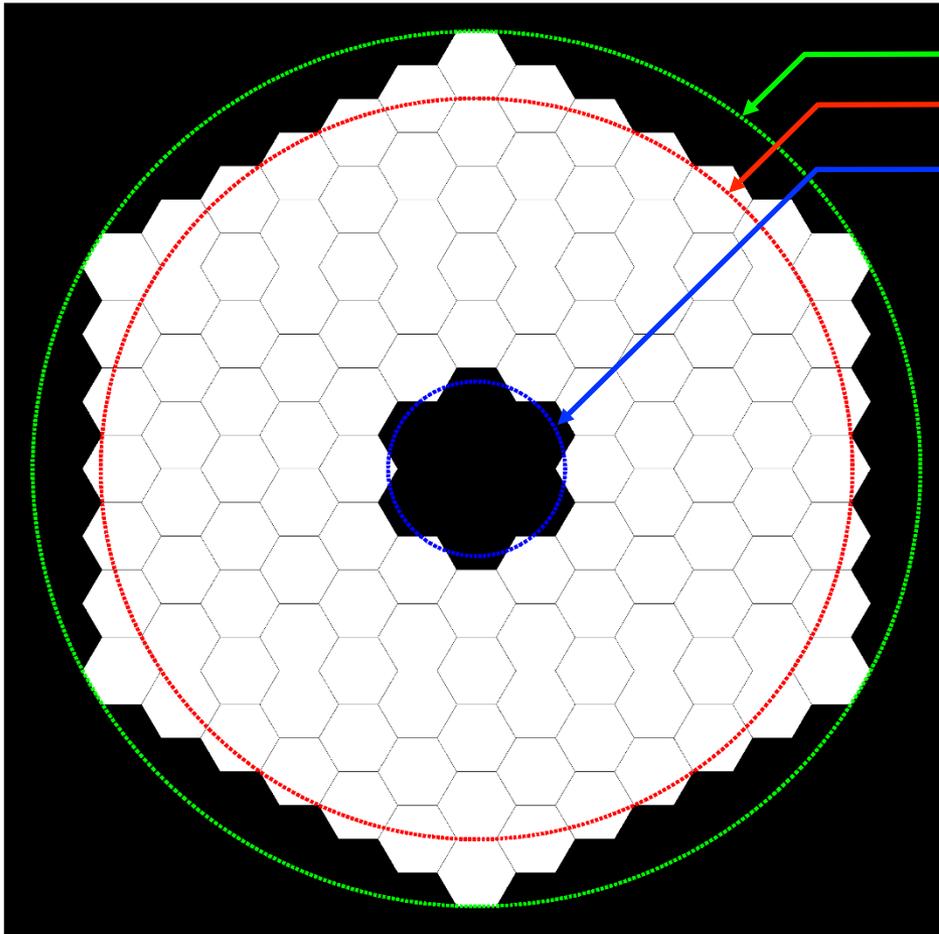
LUVOIR “A” Science Measurement Concept

- ◎ FOV: 10 arcmin x 8 arcmin
- ◎ Wavelength:
 - 100 nm – 2.5+ μm
 - 90 nm blue cutoff stretch goal (largely dependent on coating technology development)
 - Optics & coatings should not preclude observations as red as 5.0 μm
- ◎ Diffraction-limited at 500 nm
- ◎ Spatial resolution:
 - Limiting instrument (*LUMOS*) has a spatial resolution of 30 mas for an assumed 25 μm resolution element
 - Implies a telescope focal length ≥ 172 m
- ◎ Aperture diameter:
 - Largest that can fit in an 8.4-meter x 27.4-meter fairing
 - Deemed to be 15-m by LUVOIR Engineering Team
- ◎ Operating temperature: 270 K

LUVOIR "A" Telescope Optical Design



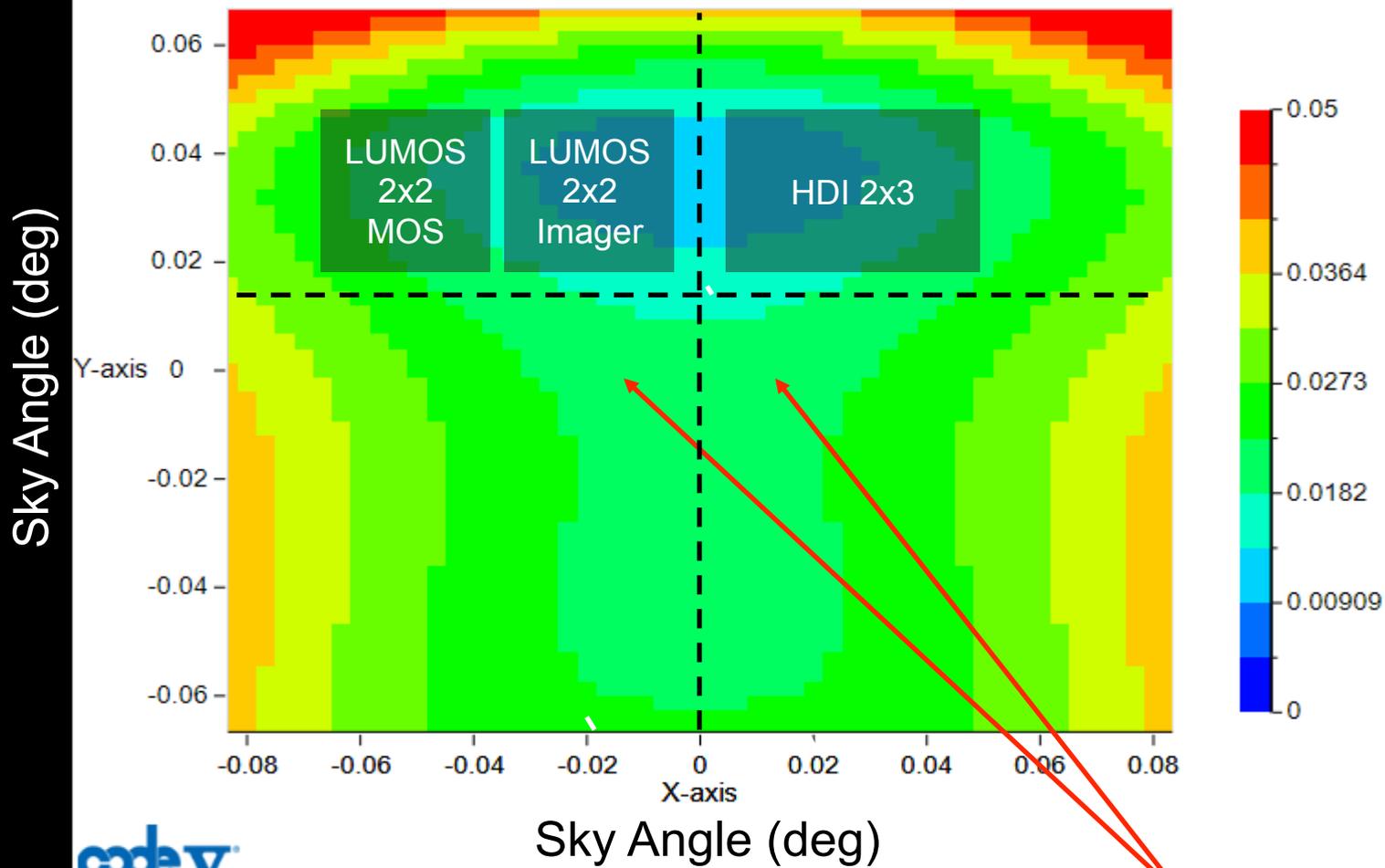
LUVOIR “A” Telescope Aperture



15.0 m
12.7 m
3.0 m

- 1.15-m flat-to-flat segments (120x)
- Central ring of array removed to accommodate Aft-optics & Secondary Mirror Obscuration
- Effective area is $\sim 135 \text{ m}^2$
- 15-m circumscribed diameter / 12.7-m inscribed diameter
- Assumes 6 mm gaps

TMA Telescope RMS WAVEFRONT ERROR (WAVES AT 1000.0 NM)

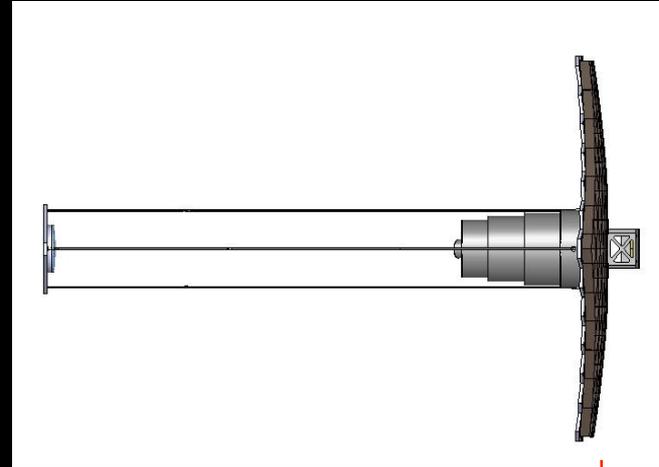
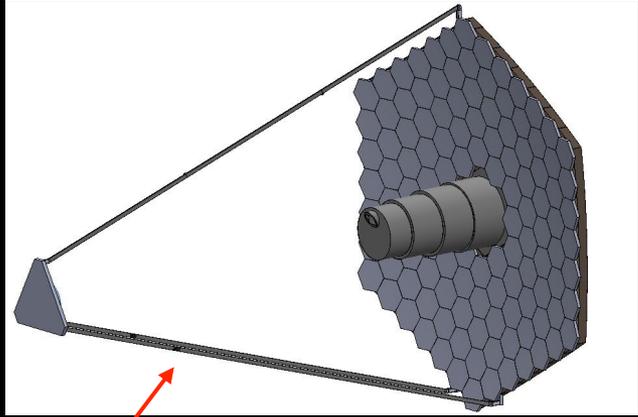


05- Jan -17
GJW
TMA Telescope

Position 1
Object Angle Field Coordinates

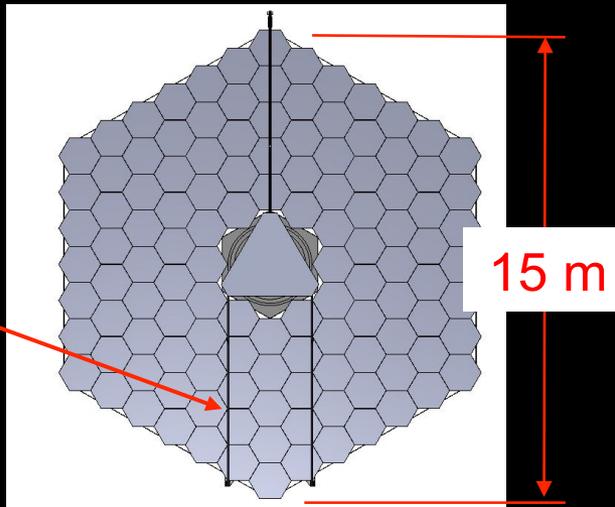
Coronagraph & CNES
Spectropolarimeter

Mechanical Design Details (1)



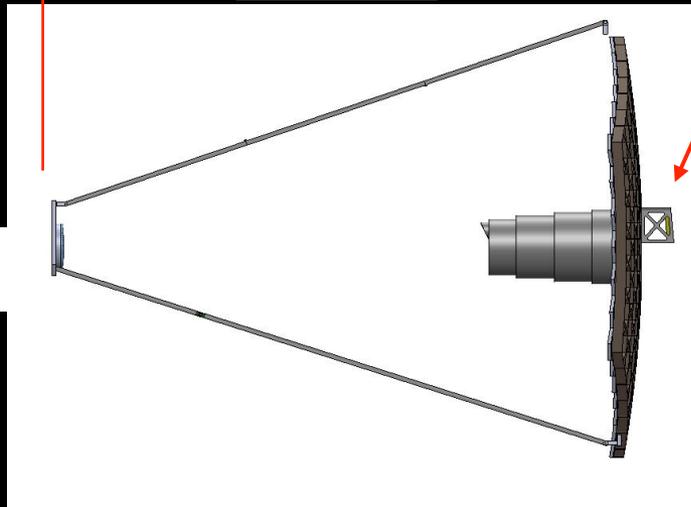
Tertiary mirror support structure

~20 m

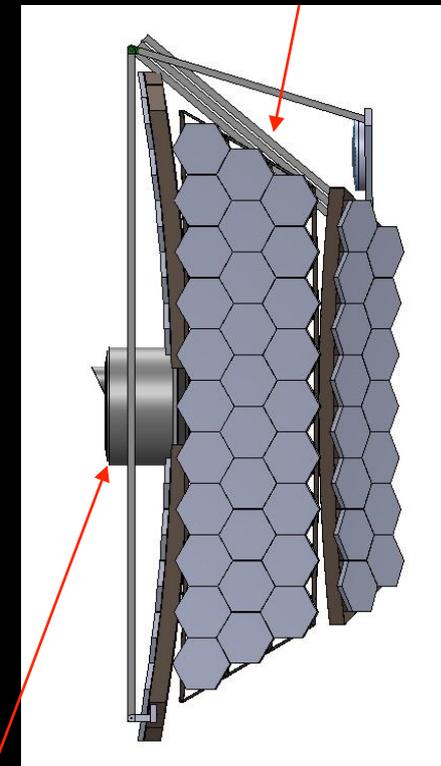
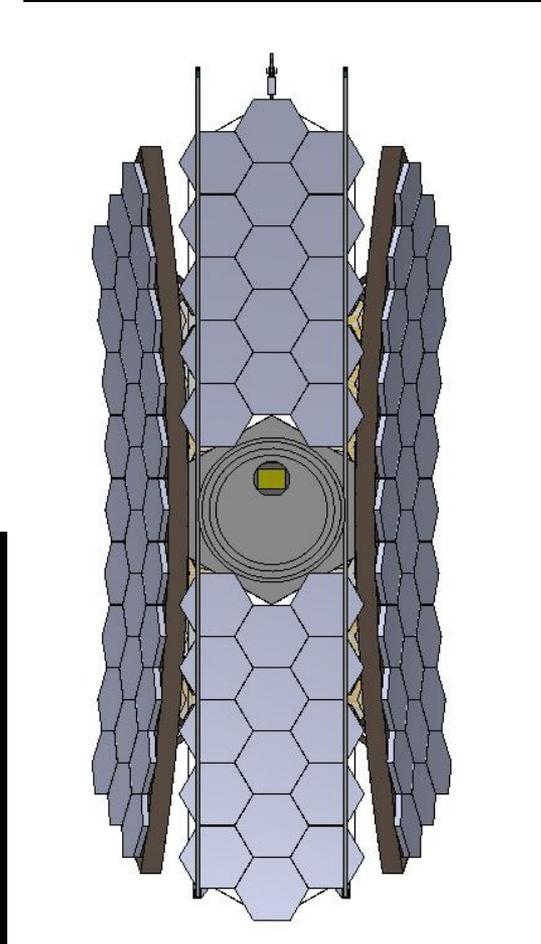
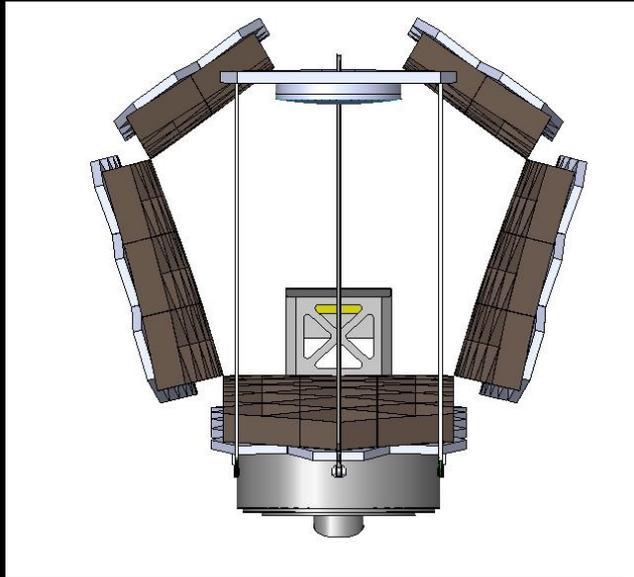


15 m

Strut cross-sections
0.05 m x 0.15 m



Mechanical Design Details (2)

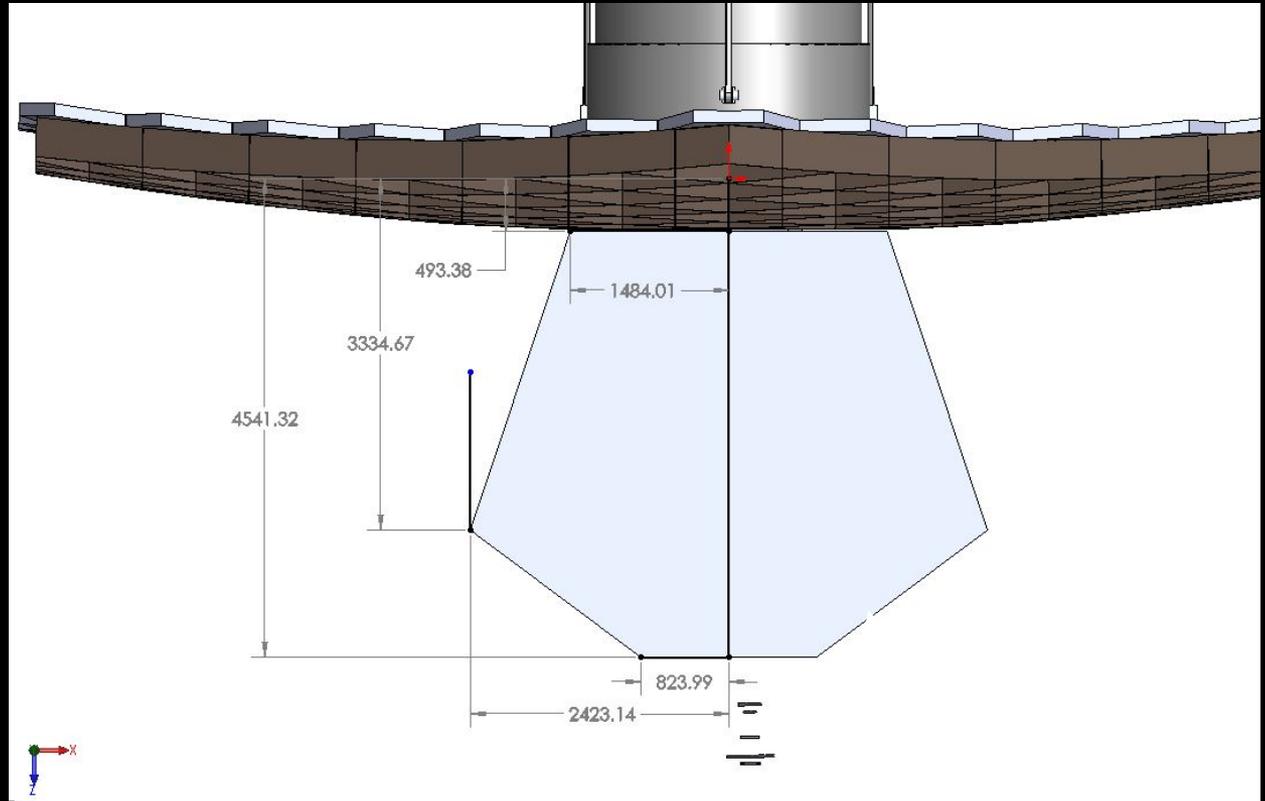
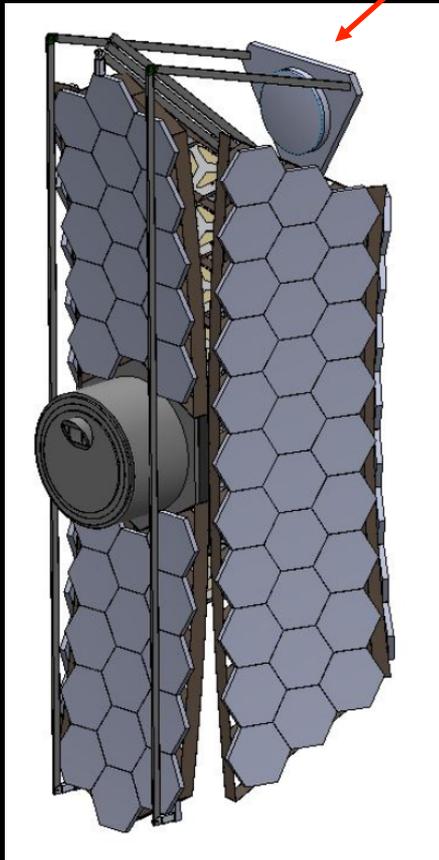


Folded
segments of
SMSS

Stowed Telescoping Boom is 1.5 m deep

Mechanical Design Details (3)

Secondary Mirror

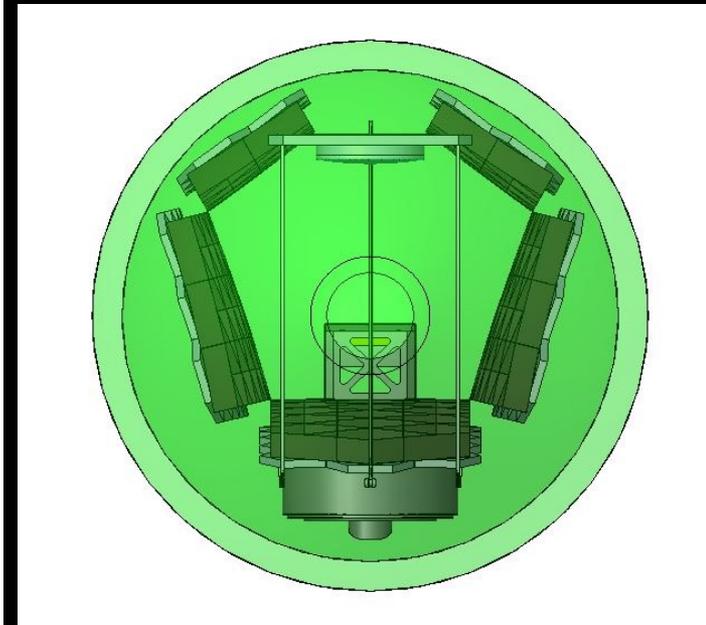
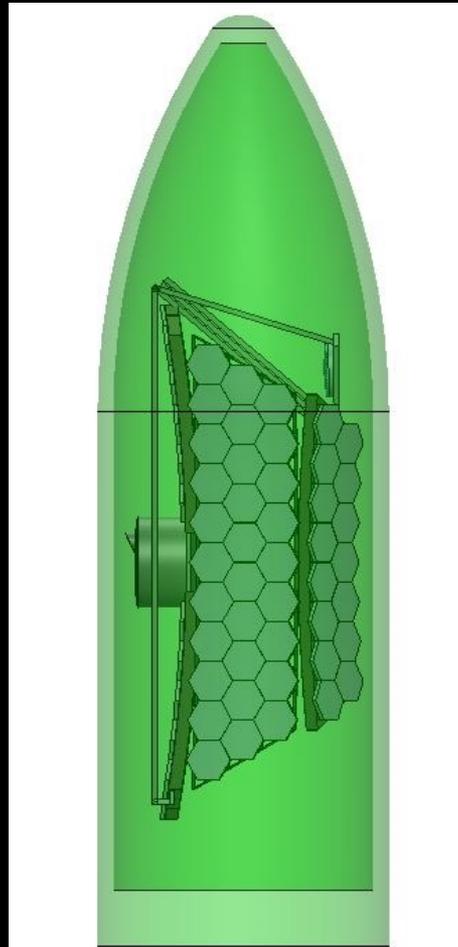


Footprint for Backplane Support Frame (BSF) and Instruments, given OTE stowed arrangement

Mechanical Design Details (4)

Not Shown:

- Sunshield and deployment system
- Spacecraft bus
- Primary Mirror "frill"



6' tall
Fairing
Guard

Mechanical Design Details (5)

Deployed Boom

Fine Steering
Mirror Assembly

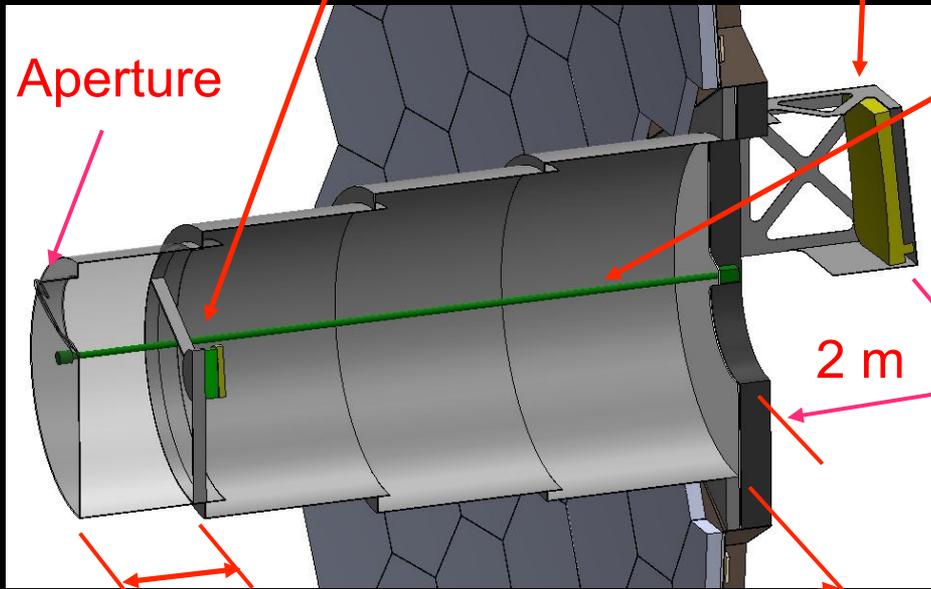
Tertiary Mirror

Aperture

2 m

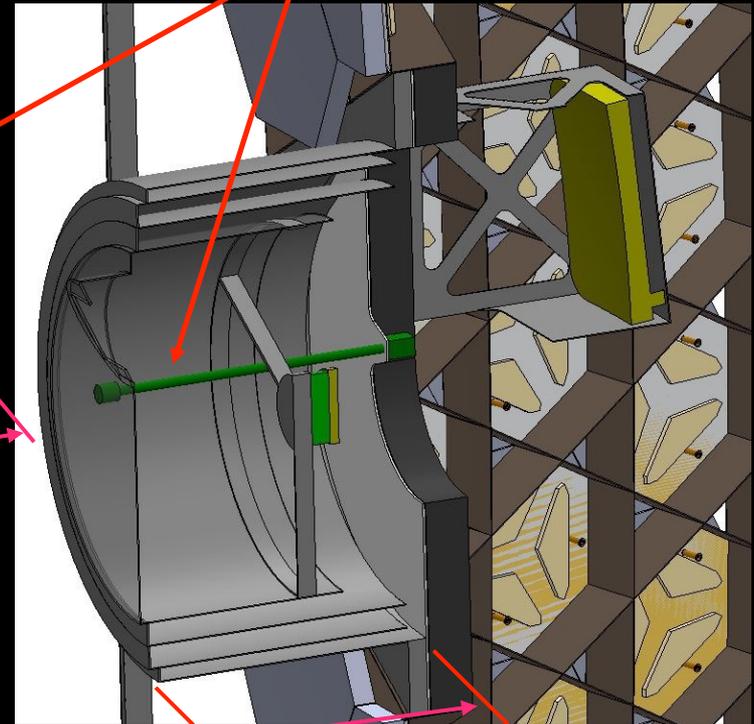
1.3 m

5 m



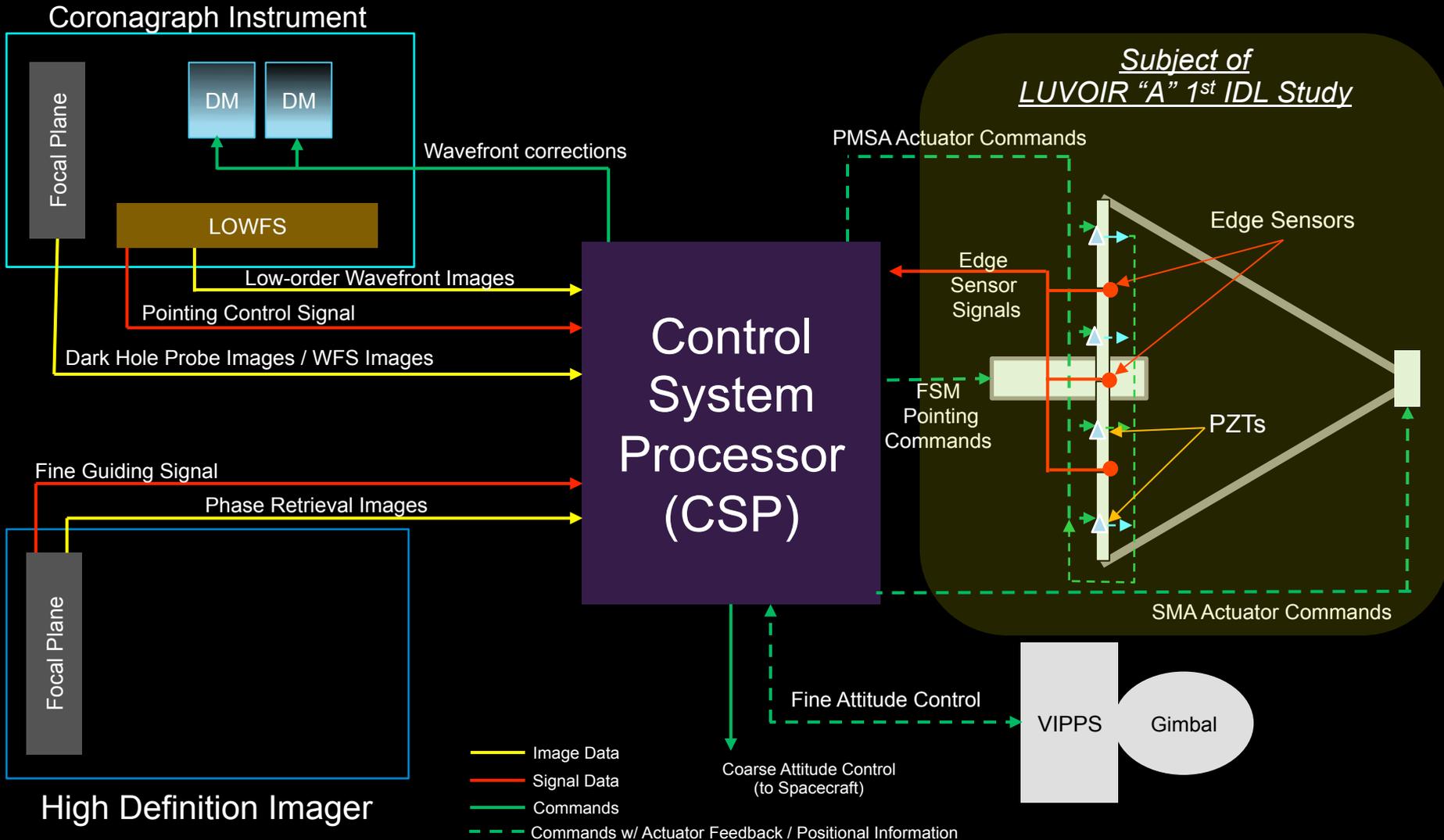
Stowed Boom

STEM deployers
(retractable if desired)



1.5 m

Control System Block Diagram (So Far)



Concept for PM Segment Phasing (1)

- Introduced a closed-loop control system at the primary mirror segments to maintain segment-to-segment phasing
- Edge sensors on each segment measure picometer-level rigid-body motions at high speeds
- In response, piezoelectric (PZT) actuators move PM segments at picometer level
- Closed-loop system creates a “virtual monolith”

Concept for PM Segment Phasing (2)

Edge Sensors

- ◎ Capacitive edge sensors
 - Two sensors per edge shared between segments: 622 sensors total
 - Provide 6-degree of freedom motion of each segment
- ◎ Similar sensors are being developed for ground-based systems (TMT, EELT, GMT) and have been used on Keck
- ◎ Challenge for LUVOIR is in the electronics
 - Need high speed (~ 450 Hz) readout with picometer-level accuracy at low power
- ◎ Lab-based system has demonstrated ~ 10 pm sensing at lower speeds with custom electronics

Concept for PM Segment Phasing (3)

Segment Actuators

- ⦿ Average the 450 Hz edge sensor measurements at a rate of 5:1 to generate a 90 Hz control signal for PM segments
- ⦿ LUVOIR PM segments use exact same actuator design as JWST, except fine stage mechanical actuator is replaced with a PZT actuator
 - One PZT per actuator → 6 PZTs per PM segment for fine control of six degrees of freedom
- ⦿ On JWST, a mechanical linkage is used to “step-down” physical actuator displacement to PM segment motion
 - i.e. if actuator moves 100 microns, the mirror only moves ~1 micron
- ⦿ We will use the same linkage for the PZTs such that a 0.250 nm PZT step (which is easy) corresponds to ~2 pm motion of the mirror segment

Priority Telescope “To-Do”

- ◎ Add mechanical design fidelity to:
 - Primary mirror backplane
 - Primary mirror segment mechanical structure (heater, whiffles, delta frame, actuators, mounting points)
- ◎ Perform dynamic stability analysis
 - For launch loads on stowed configuration
 - For jitter disturbance on deployed configuration
- ◎ Re-visit thermal control system
 - Incorporating new backplane wing sections
 - Incorporating actuator drive electronics in each primary mirror segment assembly
- ◎ Add fidelity to mechanisms
 - Launch locks, deployment motors, latches, snubbers, etc.
- ◎ Perform straylight analysis and size baffling

High Definition Imager Design

February 21, 2017

HDI Technical Overview (1/2)

- ◎ Two-channel Imaging Instrument:
 - UV/Vis Imaging (200 nm - $\sim 1.0 \mu\text{m}$)
 - Diffraction-limited performance at 500 nm
 - Nyquist sampled at 400 nm
 - NIR Imaging ($\sim 1.0 \mu\text{m} - 2.5 \mu\text{m}$)
 - Diffraction-limited performance at 1.2 μm
 - Nyquist sampled at 1.2 μm
- ◎ Each channel will contain a suite of spectral filters:
 - Narrow (R $\sim 50-100$)
 - Medium (R $\sim 20-40$)
 - Broadband (R $\sim 3-5$)
 - At least one slitless grism/prism option with R $\sim 200-500$
- ◎ Field-of-view: 2 x 3 arcmin
 - Both channels view the same patch of sky

HDI Technical Overview (2/2)

⦿ Exposure times:

- For most extragalactic sources and stellar population observations:
 - Total observation times of up to 200 hrs.
 - Composed of many exposures of 500-1000 s each
- High-speed photometry will require exposures of 50 – 100 ms
 - Will only be required over a small area of the focal plane array (perhaps a single SCA of the entire FPA)

⦿ Dynamic Range:

- Desire the ability to define a region of the focal plane with reduced sensitivity (or faster readout) for both astrometry and solar system observations

HDI Special Modes :

- ⦿ High-Precision Astrometry (for measuring exoplanet mass)
 - Astrometric precision of $< 5 \times 10^{-4}$ pixels
 - Requires a Pixel Calibration System to calibrate pixel geometry
- ⦿ Fine-guiding
 - HDI is the primary fine-guidance sensor for the LUVOIR observatory
 - Similar to WFIRST operation
 - Requires ability to define regions of focal plane with faster readout
 - Should have capability in both UV/Vis and NIR channels
- ⦿ Image-based Wavefront Sensing (i.e. phase retrieval) for telescope commissioning and maintenance
 - Similar to role played by NIRCcam on JWST
 - Requires inclusion of:
 - Weak-lenses for generating defocused images
 - Dispersed Hartmann Sensor (DHS) gratings for coarse piston sensing
 - Pupil Imaging Lens (PIL) subsystem

HDI Detector Concept – UV/Vis Channel

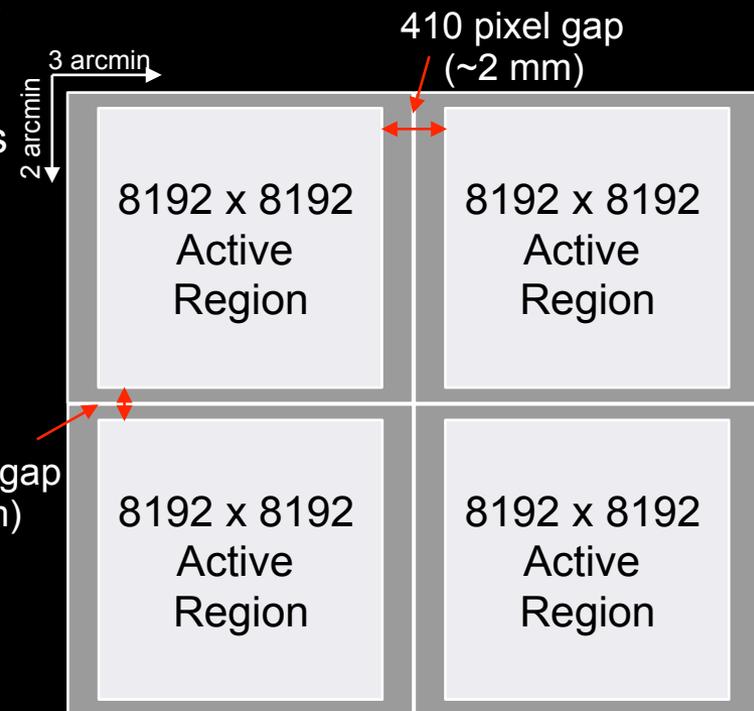
- ◉ CMOS Detector

- Pixel size = 5 μm
- Nyquist sampled at 400 nm
 - Defined as: 1 pixel = $\lambda / (2 \cdot D)$
 - $\lambda = 400 \text{ nm}$; $D = 15.08 \text{ m}$; $\diamond 1 \text{ pixel} = 2.74 \text{ mas}$
- Read noise: $\sim 2.5 \text{ e}^-$
- Dark Current: Assume $0.001 \text{ e}^-/\text{pix}/\text{s}$
- Assume same QE as WFC3 UVIS CCD detector
- Operating temperature $\sim 120 \text{ K}$
- Pixel sensitivity to be stable to $\sim 1\%$ over 14 days

- ◉ For an 8k x 8k detector technology:

- Use 5 x 8 tiling of arrays:
 - FOV = $1.90 \times 3.12 \text{ arcmin}$
 - $40,960 \times 65,536 \text{ pixels} = 2.68 \text{ Gpix}$
 - $209 \times 342 \text{ mm}$ focal plane array (including gaps)
- Gaps are as shown at right: 205 pixel gap ($\sim 1 \text{ mm}$)

- ◉ Assume 16 bits/pixel: 5.4 Gbytes per image



HDI Detector Concept – NIR Channel

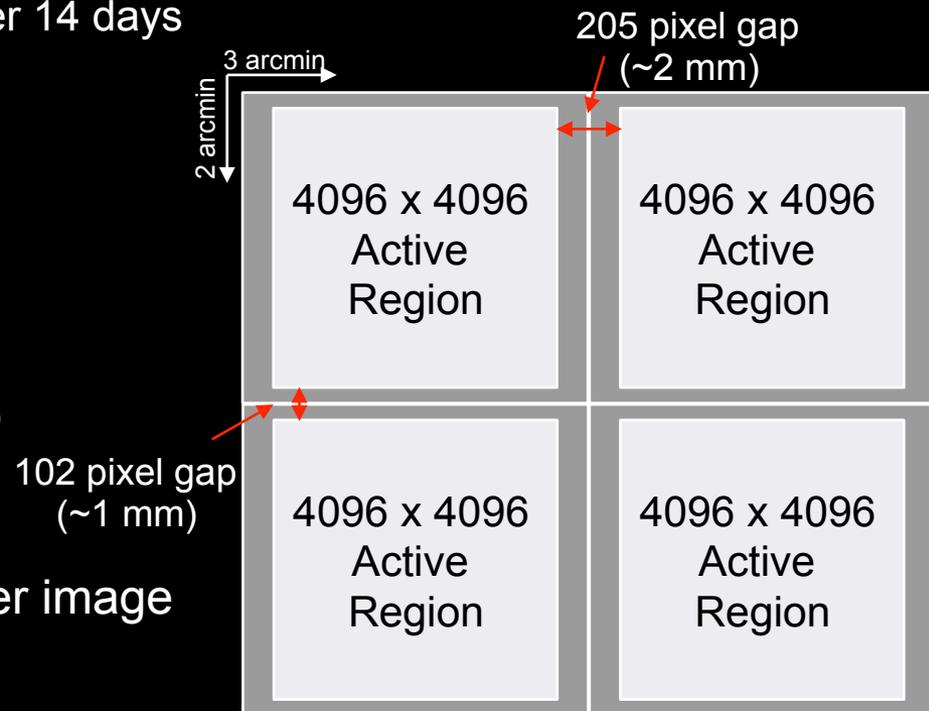
○ H4RG Detector

- Pixel size = 10 μm
- Nyquist sampled at 1200 nm
 - Defined as: 1 pixel = $\lambda / (2 \cdot D)$
 - $\lambda = 1200 \text{ nm}$; $D = 15.08 \text{ m}$; \diamond 1 pixel = 8.2 mas
- Read noise, dark current, QE adopted from WFIRST H4RG specs
- Operating temperature $\sim 70 \text{ K}$
- Pixel sensitivity to be stable to $\sim 1\%$ over 14 days

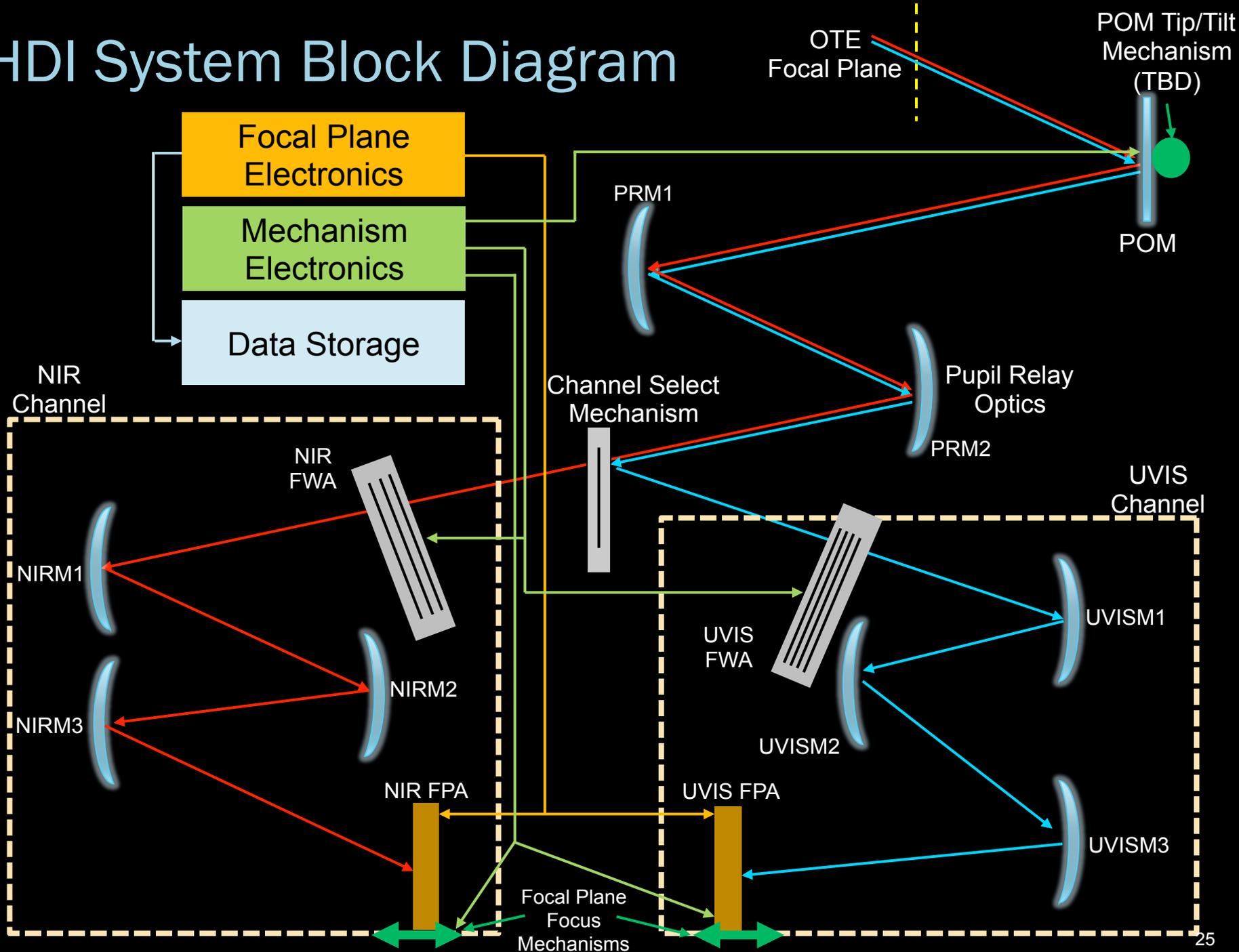
○ For an 4k x 4k detector technology:

- Use 4 x 5 tiling of arrays:
 - FOV = 2.28 x 2.91 arcmin
 - 16,384 x 20,480 pixels = 335 Mpix
 - 167 x 213 mm focal plane array (including gaps)
- Gaps are as shown at right:

○ Assume 16 bits/pixel: 0.671 Gbytes per image



HDI System Block Diagram



HDI Thermal Design

- ◎ Three thermal zones within the instrument using passive cooling:
 - 260 K
 - Instrument housing
 - Pupil relay optics
 - UVIS channel optics
 - 120 K
 - NIR channel optics
 - UVIS focal plane
 - 70 K
 - NIR focal plane

Priority HDI “To-Do”

- ◎ Resolve a small volume allocation violation
 - A few millimeters of the NIR channel thermal shroud encroaches beyond the allocated instrument volume
- ◎ Finalize optical design
 - Grisms, weak lenses, pupil imaging lens, etc.
 - Investigate optimizing the design for UVIS throughput
- ◎ Re-visit thermal design and radiator sizing for the three thermal zones
 - 120 K may be colder than is needed for UVIS detector
- ◎ Re-visit number of elements and element type in the channel select mechanism